



International Space Station (ISS) Measured Vibratory Environment

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Outline



- A Couple of Important Notes on Vibratory Data
- · Vehicle
 - Shuttle Docking
 - Air conditioner [SKV]; Система Кондиционирования Воздуха (СКВ)
 - Control Moment Gyroscope (CMG)
- Experiment
 - Gas Analysis System for Metabolic Analysis of Physiology (GASMAP)
 - Microencapsulation Electrostatic Processing System (MEPS)

• Crew

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- Public Affairs Office (PAO) Event
- Exercise
 - Resistive Exercise Device (RED)
 - Cycle Ergometer with Vibration Isolation System (CEVIS)
 - Russian Velosiped (Velo)
 - Russian Segment Treadmill Vibration Isolation System (TVIS)
- Principal Component Spectral Analysis (PCSA)





- Always demean vibratory data because Space Acceleration Measurement System (SAMS) sensors are not intended for low-frequency, quasi-steady measurements. If acceleration measurements below 0.01 Hz are needed, then seek Microgravity Acceleration Measurement System (MAMS) data. This instrument is dedicated to measurement below 0.01 Hz with special procedures and analysis to get precise accounting of quasi-steady regime, including the mean.
- Vibratory acceleration data are not mathematically mapped to other locations because the rigid body assumption does not hold at these higher frequencies.

Shuttle Docking QUALIFY



Date last modified 12/30/02

Data Description	
Sensor	SAMS, 121f02 250.0 sa/sec (100.00 Hz)
Location	LAB1O2, ER1, Drawer 1
Inc/Flight	Increment: 5, Flight: UF2
Plot Type	spectrogram
 Notes: The Shuttle docks at the forward end of the US Lab to a Pressurized Mating Adapter (PMA-2). Initial contact is referred to as "softmate" even though its impact is typically greater in magnitude than the "hardmate" event. A typical Shuttle docking is as follows: (1) initial contact and capture (softmate) (2) pause several minutes to let relative motion between the two spacecraft damp out (3) drive latches to make solid mechanical connection (hardmate) Upon completion of the hardmate, the two spacecraft effectively become one structure from a vibratory acceleration transmission perspective. This is usually evidenced by measurements made on the space station, which show the signature of the Shuttle's Ku-band antenna. This antenna is nearly continuously dithered at 17 Hz to prevent mechanical stiction and usually is accompanied by higher harmonics (most notably 34 and 51 Hz). 	
Regime.	Vibratory
Category	Vehicle
Source:	Shuttle Docking
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Shuttle Docking QUANTIFY



Data Description



Data Description		
Sensor	SAMS, 121f02 250.0 sa/sec (100.00 Hz)	
Location	LAB1O2, ER1, Drawer 1	
Inc/Flight	Increment: 4, Flight: 8A	
Plot Type	spectrogram	

Notes:

- This Russian air conditioner is part of the Environmental Control and Life Support System (ECLSS).
- As part of preparation for an ExtraVehicular Activity (EVA), the crew will move ducting in order to work with hatches. It has been noted that this is accompanied by a temporary deactivation of the air conditioner.
- Nominally one SKV is operating, but there are times when both SKV-1 and SKV-2 are on.
- The nearly continuous operation of this equipment produces a narrowband disturbance near 23.5 Hz. This spectral peak serves as vibratory beacon signal.

Regime:	Vibratory
Category:	Vehicle
Source:	SKV Air Conditioner

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Data Description		
Sensor	SAMS, 121f02 250.0 sa/sec (100.00 Hz)	
Location	LAB1O2, ER1, Drawer 1	
Inc/Flight	Increment: 4, Flight: 8A	
Plot Type	interval RMS	

Notes:

For the 8-hour period shown in the figure, Parseval's theorem was used to derive the RMS acceleration level for a narrow band around the SKV air conditioner's operating frequency. The results for the frequency band from 23 to 24 Hz is tabulated below:

4	SKV	GMT	Median (µg _{RMS})
	ON	08:00 - 11:00	55.4
É	OFF	11:00 - 13:28	6.8
1	ON	13:28 - 16:00	91.2
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Regime:	Vibratory
Category:	Vehicle
Source:	SKV Air Conditioner

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Control Moment Gyroscope (CMG) QUALIFY



Data Description		
Sensor	SAMS, 121f03 500.0 sa/sec (200.00 Hz)	
Location	LAB101, ER2, Lower Z Panel	
Inc/Flight	Increment: 5, Flight: UF2	
Plot Type	PCSA	

Notes:

There are 4 control moment gyroscopes (CMGs) located on the Z1 truss structure (currently, only 3 are operational). These rotate at 6,600 revolutions per minute (RPM) within 1 RPM (1/60 Hz) to provide non-propulsive attitude control for the space station. The principal component spectral analysis (PCSA) plot at the left summarizes the acceleration spectrum for a 3-day span. As seen by the narrow spectral peak at 110 Hz, these gyros are tightly controlled in frequency.

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Regime:	Vibratory
Category:	Vehicle
Source:	Control Moment Gyroscope (CMG)

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Control Moment Gyroscope (CMG) Data Description OUANTIFY SAMS, 121f03 Sensor 500.0 sa/sec (200.00 Hz) sams2, 121f03 at LAB1O1, ER2, Lower Z Panel:[191.54 -40.54 135.25] Increment: 5, Flight: UF2 500.0 sa/sec (200.00 Hz) $\Delta f = 0.122$ Hz, Nfft = 4096 CMGs, 109.983 < f < 110.017 Hz Sum Hanning, k = 31141 Span = 72.00 hours Location LAB101, ER2, Lower Z Panel Temp. Res. = 8.192 sec, No = 0 Start GMT 05-October-2002, 278/00:00:00 1500 Inc/Flight Increment: 5, Flight: UF2 Plot Type interval RMS 1381 Notes: 1274 The plot shows interval RMS values during a 3day period for a narrow frequency band (109.983 to 110.071 Hz) around the CMG operating frequency (110 Hz). Statistics gathered for this time frame show: 95th percentile: 94.9 μg_{RMS} RMS Acceleration (µg_{RMS}) 963 median: $47.7 \,\mu g_{RMS}$ $50.9 \ \mu g_{RMS}$ mean: This plot also shows that this small frequency band will also register much higher RMS levels not necessarily attributable to nominal operation of the 647 CMGs. 364 150 100 50 00:00 04:00 08:00 12:00 16:00 20:00 00:00 04:00 08:00 12:00 16:00 20:00 00:00 04:00 08:00 12:00 16:00 20:00 00:00 GMT 05-October-2002, 278/hh:mm from: t/pub/pad/, \$Name: pop3_06-17-2002 \$, 24-Jun-2002,06:19:13.602 Vibratory Regime: Category: Vehicle



Source:	Control Moment Gyroscope (CMG)

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Gas Analysis System for Metabolic Analysis of Physiology (GASMAP) Qualify



Date last modified 10/2/02

Data Description		
Sensor	SAMS, 121f02 500.0 sa/sec (200.00 Hz)	
Location	LAB1O2, ER1, Drawer 1	
Inc/Flight	Increment: 4, Flight: UF1	
Plot Type	spectrogram	

Notes:

The GASMAP device is used to monitor and analyze both inhaled and exhaled breath streams to determine their gas concentrations. Two GASMAP signatures are noted in the figure near 60 Hz: a sample pump and a fan. This equipment is located in the HRF rack #1 (LAB1S2) and used for the Pulmonary Function in Flight (PuFF) experiment, which studies effects of extravehicular activity (EVA) and long-term exposure to microgravity on the pulmonary function. As discussed with payload developers, the frequency changes away from nominal operation are likely a function of varying loads on the sample pump. The pump frequency ranged from 54 Hz (for brief calibrations) up to 62 Hz after the second PuFF run. The fan is tightly controlled in frequency but also transitions from baseline of about 57.4 Hz up to 58.1 Hz noted during calibrations.

Regime:	Vibratory
Category:	Experiment Equipment
Source:	GASMAP

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Microencapsulation Electrostatic Processing System (MEPS) Qualify



Date last modified 10/2/02

Data Description		
Sensor	SAMS, 121f02 250.0 sa/sec (100.00 Hz)	
Location	LAB1O2, ER1, Drawer 1	
Inc/Flight	Increment: 5, Flight: UF2	
Plot Type	spectrogram	

Notes:

The MEPS is located in locker #8 of ER1 (LAB1O2) near the 121f02 sensor in RTS drawer 1. This 24-hour spectrogram readily shows 5 MEPS sample runs with the start/stop delimited by 2 narrowband signals. The weaker of these was centered at 47.1 Hz and the stronger at 52.8 Hz. Note that each of the 5 runs is preceded by the impulsive events of Process Chamber Module (PCM) insertion.

K	Regime:	Vibratory	
X	Category:	Experiment Equipment	
	Source:	MEPS	

Microencapsulation Electrostatic Processing System (MEPS) Quantify



Data Description	
Sensor	SAMS, 121f02 250.0 sa/sec (100.00 Hz)
Location	LAB1O2, ER1, Drawer 1
Inc/Flight	Increment: 5, Flight: UF2
Plot Type	interval RMS

Notes:

This interval RMS versus time plot corresponds to the same time frame as that of the 24-hour spectrogram shown on the "qualify" page. This figure shows the contribution of the 2 narrowband signals that accompany the MEPS runs. The red trace gives the RMS acceleration for the fainter, low-frequency signal centered at 47.1 Hz. This signal steps from about 14 to 24 ug_{RMS} in the frequency range from 45.85 to 47.39 Hz. Likewise, the blue trace for the stronger signal centered at 52.8 Hz steps from about 9 to 54 ug_{RMS} in the frequency range from 52.63 to 53.07 Hz. The largest transient in this time frame was 103 mg and is attributed to insertion of the Process Chamber Module (PCM) for the 2nd sample run.

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	Regime:	Vibratory
	Category:	Experiment Equipment
	Source:	MEPS

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Sleep/Wake **QUALIFY**



Data Description		
Sensor	SAMS, 121f03 62.5 sa/sec (25.00 Hz)	
Location	LAB101, ER2, Lower Z Panel	
Inc/Flight	Increment: 5, Flight: UF2	
Plot Type	spectrogram	

Notes:

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This figure shows 3 sleep periods over a 64-hour span. The impact of crew wake periods relative to sleep is primarily below about 6 Hz. This is seen as a shift toward the blue end of the PSD magnitude color scale below about 6 Hz during the 3 sleep periods. The transition from sleep to wake is typically a sudden event owing to a wake alarm, while the transition from wake to sleep is gradual as might be expected. Signatures for both Russian air conditioners (SKV-1 and SKV-2) are also seen here toward the top of this figure at about 23.3 Hz. The slightly lower frequency and more intense SKV is on for this entire 64-hour duration, while the other one starts just after the end of the 2^{nd} sleep period.

Regime:	Vibratory
Category:	Crew
Source:	Sleep/Wake

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Sleep/Wake QUANTIFY



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Data Description

Public Affairs Office (PAO) Event QUALIFY



Data Description		
Sensor	SAMS, 121f02 250.0 sa/sec (100.00 Hz)	
Location	LAB1O2, ER1, Drawer 1	
Inc/Flight	Increment: 5, Flight: UF2	
Plot Type	spectrogram	

Notes:

For a microgravity environment that mimics crew sleep, one can consider PAO events that occupy the entire crew (note that some PAO events occupy only part of the crew). These events typically last at least 10 minutes or so with the crew gathered in front of a video camera participating in an interview. During this time the crew is usually quite still, but there are times when they demonstrate various things that require them to push-off or otherwise apply a force to vehicle structure. The figure here shows the subtle contrast between nominal activity and all 3 crew participating in a PAO event from about 14:45 to about 15:00. Like crew sleep the difference lies primarily below about 6 Hz.

Regime:	Vibratory
Category:	Crew
Source:	Public Affairs Office (PAO) Event

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Public Affairs Office (PAO) Event QUANTIFY



Data Description



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Resistive Exercise Device (RED) Quantify



Data Description SAMS, 121f02 Sensor 250.0 sa/sec (100.00 Hz) LAB1O2, ER1, Drawer 1 Location Inc/Flight Increment: 5, Flight: UF2 interval RMS Plot Type

Notes:

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The interval RMS figure shown spans the entire RED exercise period and covers the frequency range below 30 Hz. This period consisted of several exercise types as annotated in the figure. Several of these types were analyzed for the frequency range below 10 Hz as tabulated below:

Туре	Maximum (µg _{RMS})	GMT Start 02-Sep-2002, 245/
baseline (no RED)	60	08:55:00
squats	301	08:58:09
heel raises	691	08:58:26
deep heel raises	336	09:02:42
deadlifts	206	09:10:46
bench/bent rows	222	09:20:21
one-leq squats	255	09:05:59
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	Regime:	Vibratory
\sim	Category:	Crew Activity
	Source:	Resistive Exercise Device (RED)

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Cycle Ergometer with Vibration Isolation System (CEVIS) **QUALIFY**



Data Description		
Sensor	SAMS, 121f05 62.5 sa/sec (25.00 Hz)	
Location	LAB1O1, ER2, Light Tray	
Inc/Flight	Increment: 4, Flight: UF1	
Plot Type	spectrogram	

Notes:

The CEVIS exercise device is located in the US Lab at LAB1P3. The narrowband peak at about 2.5 Hz marked on the lower right of the spectrogram is the pedaling frequency for this CEVIS exercise period. For Shuttle ergometer exercise, the pedaling signature was accompanied by that of shoulder sway with frequency around half the pedal rate. On the ISS for this CEVIS exercise period, the shoulder sway signature is obscured by structural modes that fall in the same frequency range.

Regime:	Vibratory
Category:	Crew
Source:	Cycle Ergometer with Vibration Isolation System (CEVIS)

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Cycle Ergometer with Vibration Isolation System (CEVIS) QUANTIFY



Data Description

Velosiped (Velo) Exercise QUALIFY



Data Description



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Frequency (Hz)

sams2, 121f03 at LAB1O1, ER2, Lower Z Panel:[191.54 -40.54 135.25]

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Increment 2 PCSA





LEGEND:

antenna dither during docked operations > SKV: air conditioner in **Russian Service Module** > ISIS: International Subrack Interface Standard drawer fans > A1: ADVASC fan > A2: ADVASC pump > A3*: ADVASC blower > A4*: ADVASC blower > A5*: ADVASC fan * frequency shift/smear > E7: EXPPCS 7th harmonic NOTE:

 \succ the axis box for this plot matches next slide

selected periods from ~ 30-May-2001 to 20-Aug-2001

see p. 43 of PIMS Inc. 2 report for more details



Increment 3 PCSA





LEGEND:

 KU: No Shuttle's Ku-band antenna dither because no docked operations
 SKV: air conditioner in Russian Service Module
 ISIS: International Subrack Interface Standard drawer fans
 >higher frequency fan overlaps vacated ADVASC (A1) fan disturbance at ~50Hz
 A1-A5: No ADVASC
 E7: No EXPPCS

NOTE:

the axis box for this plot matches previous slide

selected periods from ~ 21-Aug-2001 to 30-Aug-2001

see p. 108 of PIMS Inc. 3 report for more details



Increment 5 PCSA - Data Set #7







Increment 5 PCSA - Data Set #8



